



PennState  
College of Engineering

AEROSPACE  
ENGINEERING

# Effective Helicopter Noise Abatement Operations

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Eric Greenwood

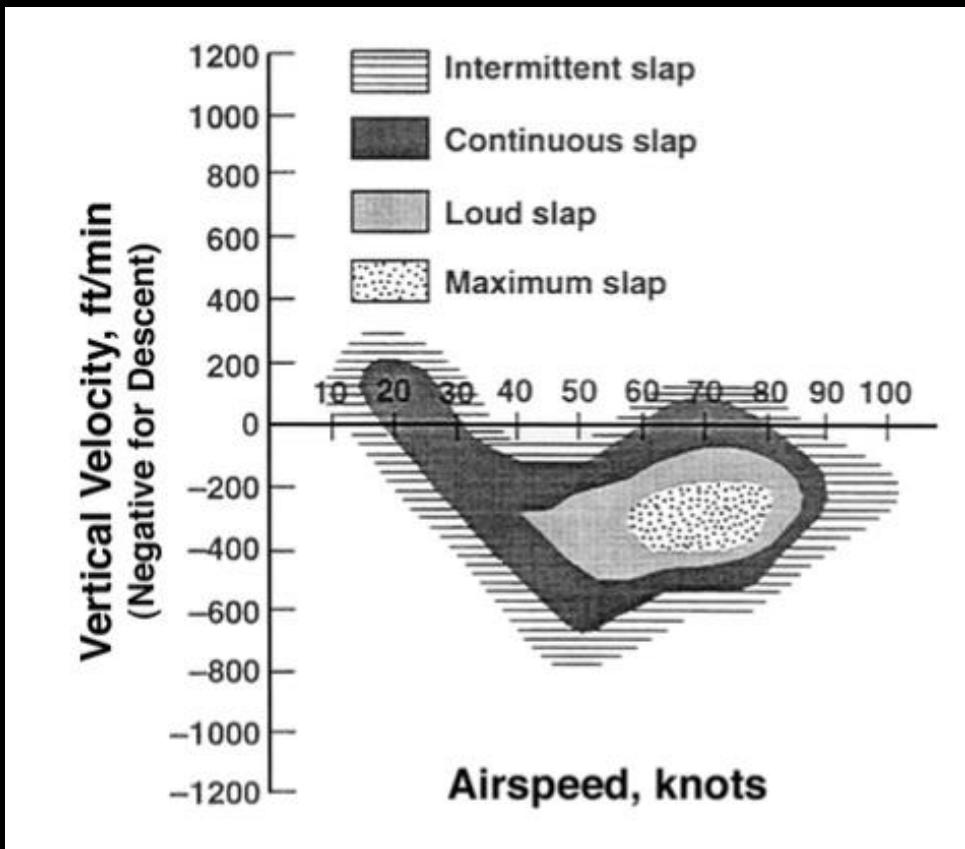
Department of Aerospace Engineering

The Pennsylvania State University





# What help do pilots get to fly quietly?



Bell 212

HAI Fly Neighborly Guide

ROBINSON  
MODEL R44

SECTION 4  
NORMAL PROCEDURES

## NOISE ABATEMENT

To improve the quality of our environment and to dissuade overly restrictive ordinances against helicopters, it is imperative that every pilot minimize noise irritation to the public. Following are several techniques which should be employed when possible.

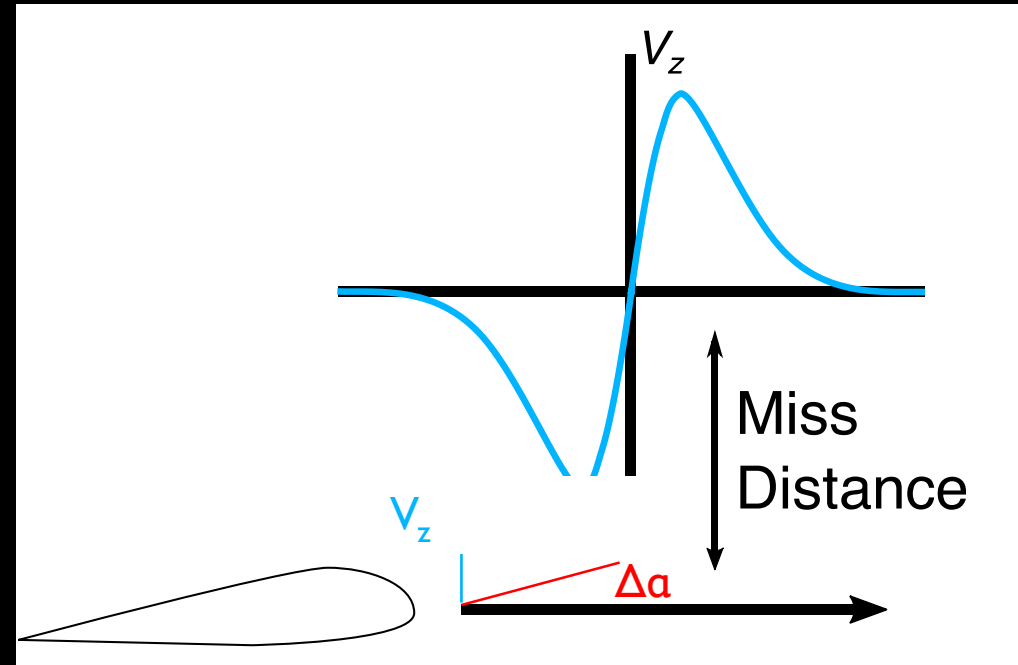
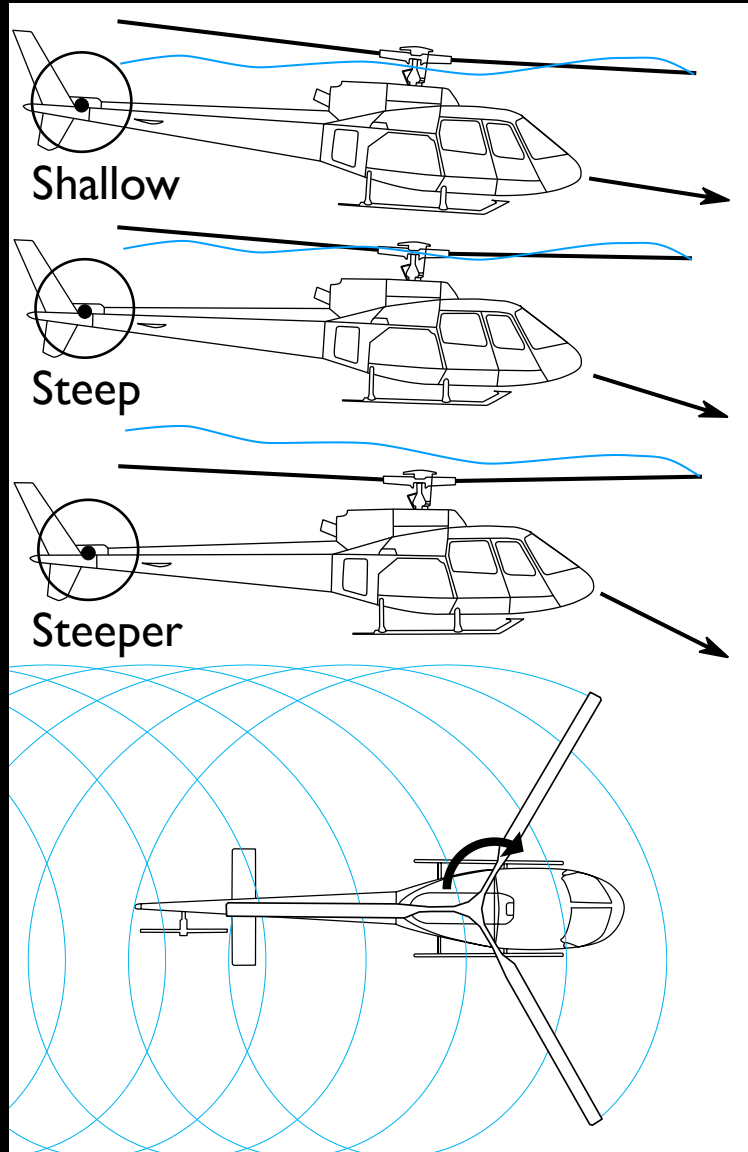
1. Avoid flying over outdoor assemblies of people. When this cannot be avoided, fly as high as practical, preferably over 2000 feet AGL.
2. Avoid blade slap. Blade slap generally occurs at airspeeds below 100 KIAS. It can usually be avoided by maintaining 100 KIAS until rate of descent is over 1000 FPM, then using a fairly steep approach until airspeed is below 65 KIAS. **With the right door vent open, the pilot can easily determine those flight conditions which produce blade slap and develop piloting techniques to eliminate or reduce it.**
3. When departing from or approaching a landing site, avoid prolonged flight over noise-sensitive areas. Always fly above 500 feet AGL and preferably above 1000 feet AGL.
4. Repetitive noise is far more irritating than a single occurrence. If you must fly over the same area more than once, vary your flight path to not overfly the same buildings each time.
5. When overflying populated areas, look ahead and select the least noise-sensitive route.

Robinson R44

Flight Manual Supplement



# “Blade slap” is usually Blade-Vortex Interaction noise



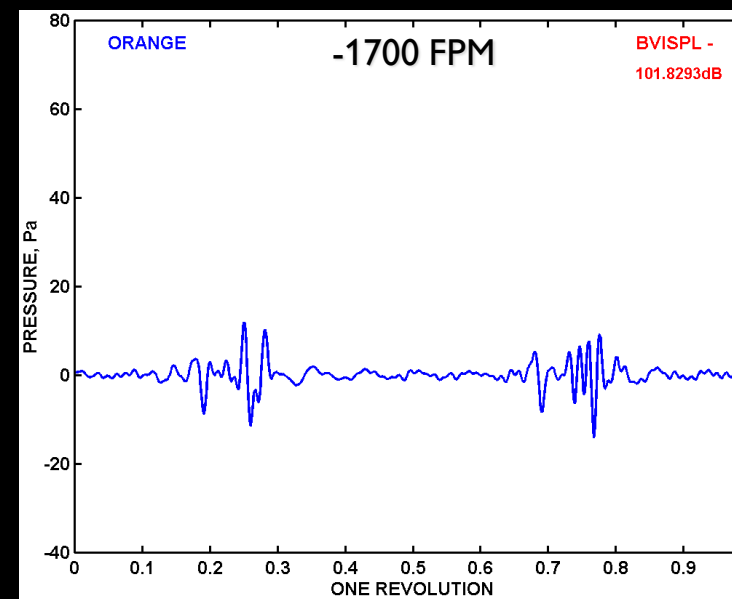
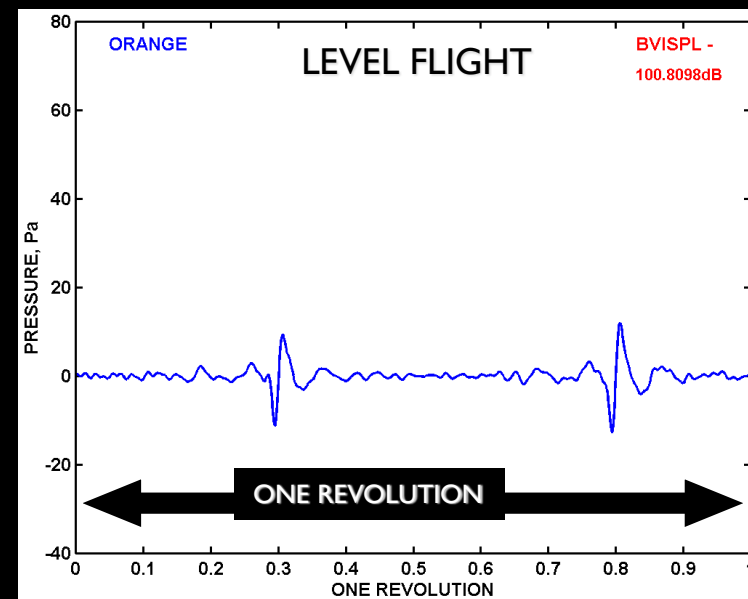
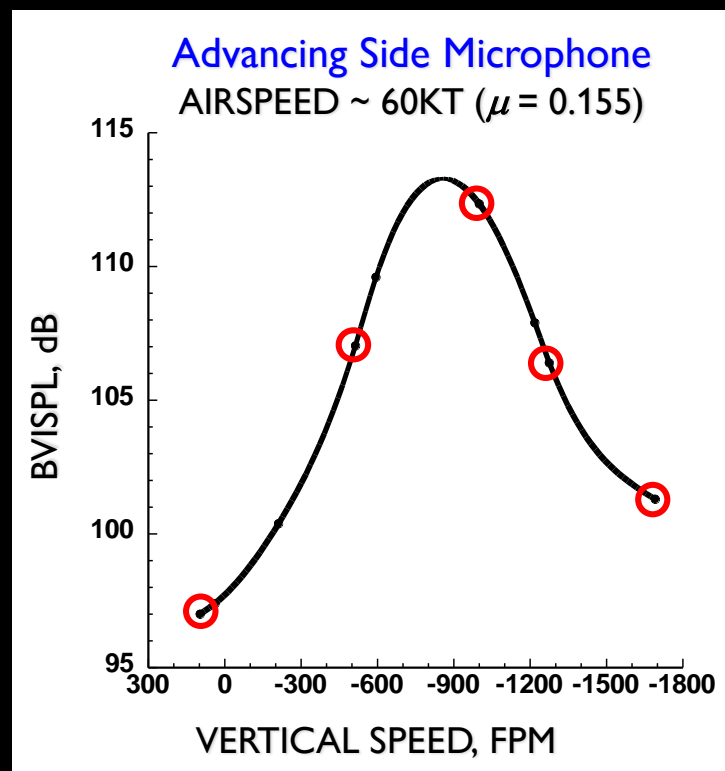
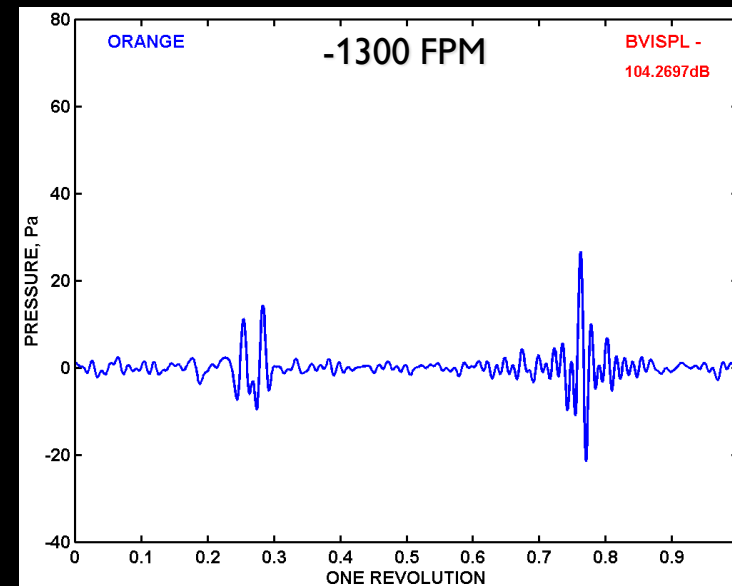
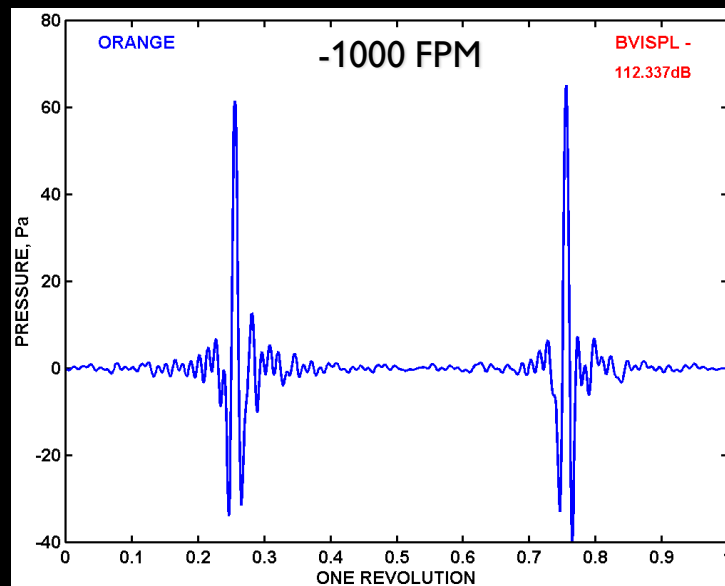
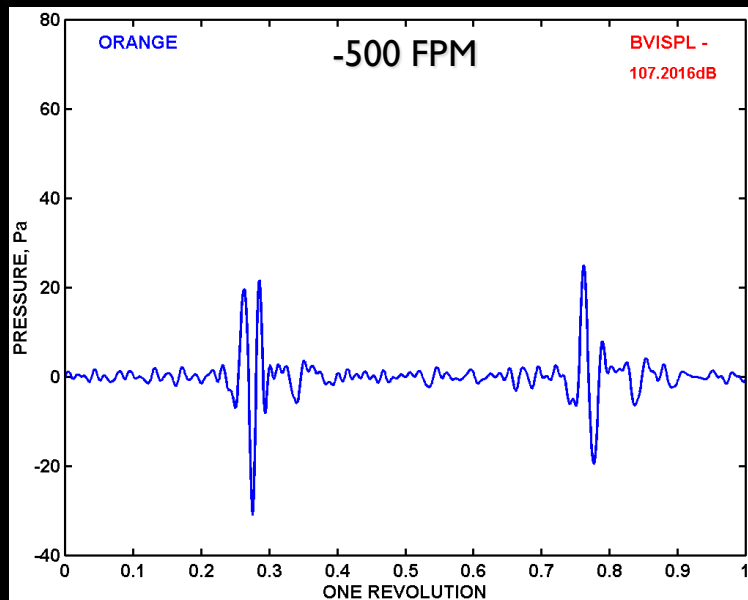
- Occurs when rotor tip vortex wake passes near rotor blades
- Rapid change in airloads results in impulsive noise
- Tends to dominate annoyance metrics when it occurs
- Magnitude and directivity strong function of rotor operating condition





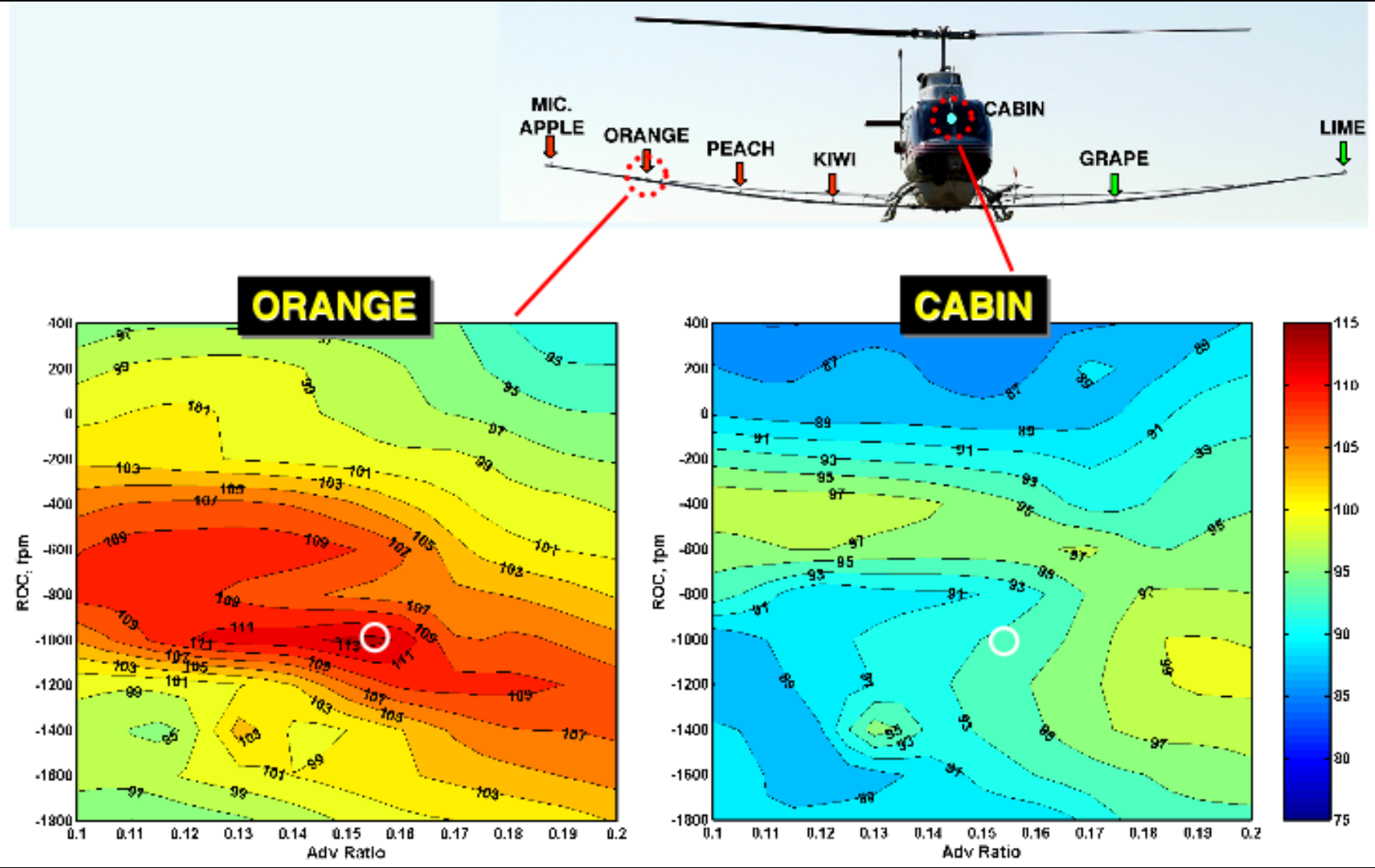
# 2006 NRTC Bell 206 Acoustic Flight Test







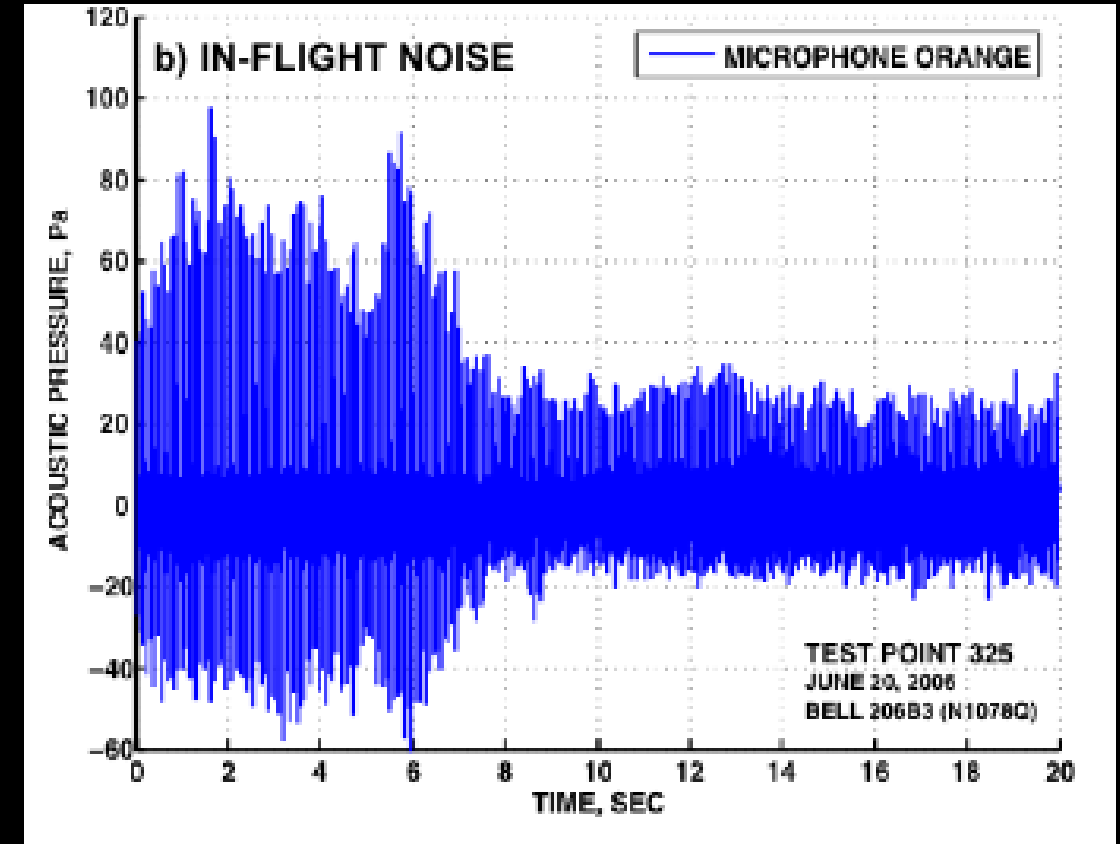
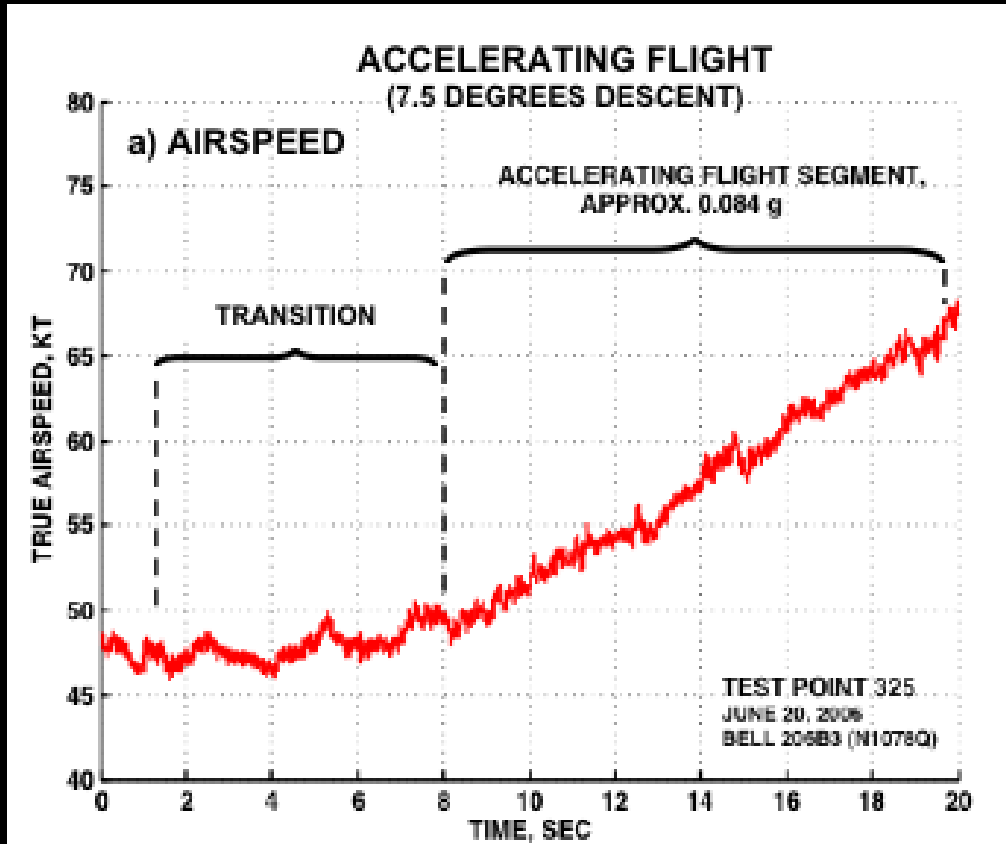
# You can't trust your ears to tell you what's happening outside the aircraft!



BVISPL, dB



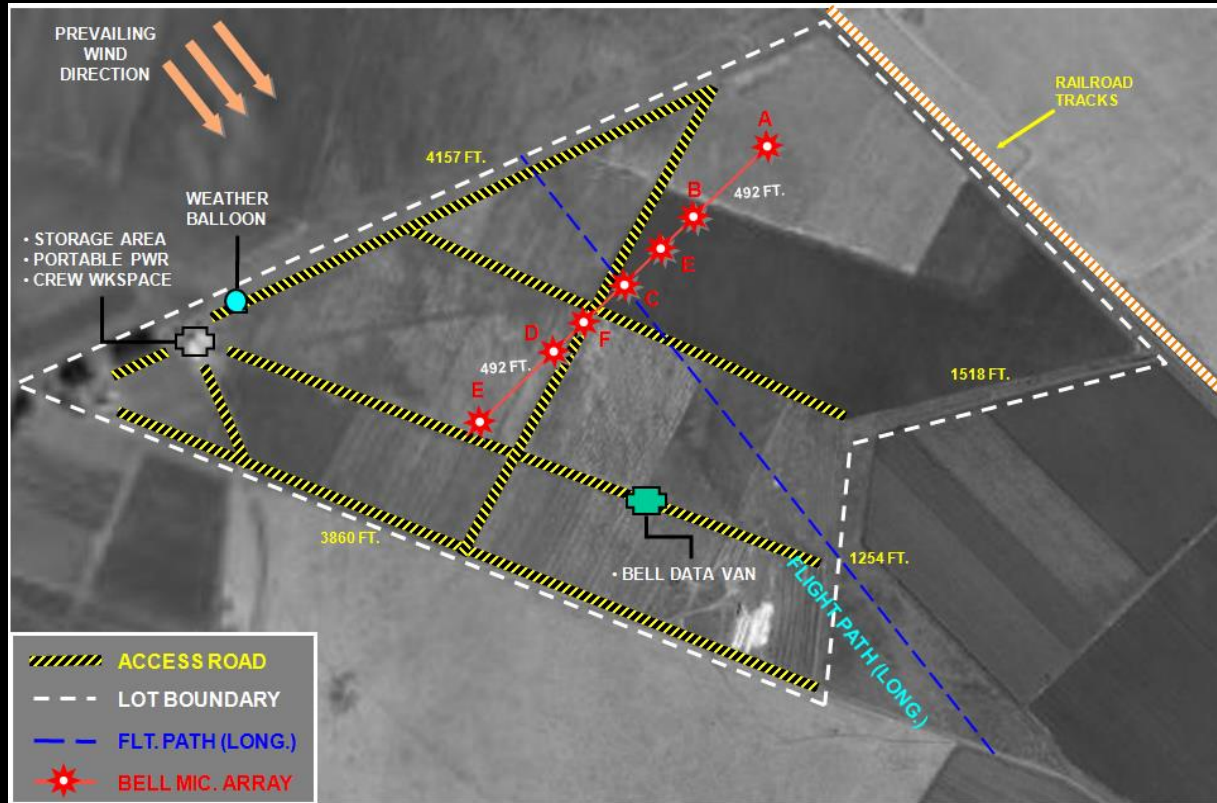
# Acceleration has a powerful influence on BVI noise







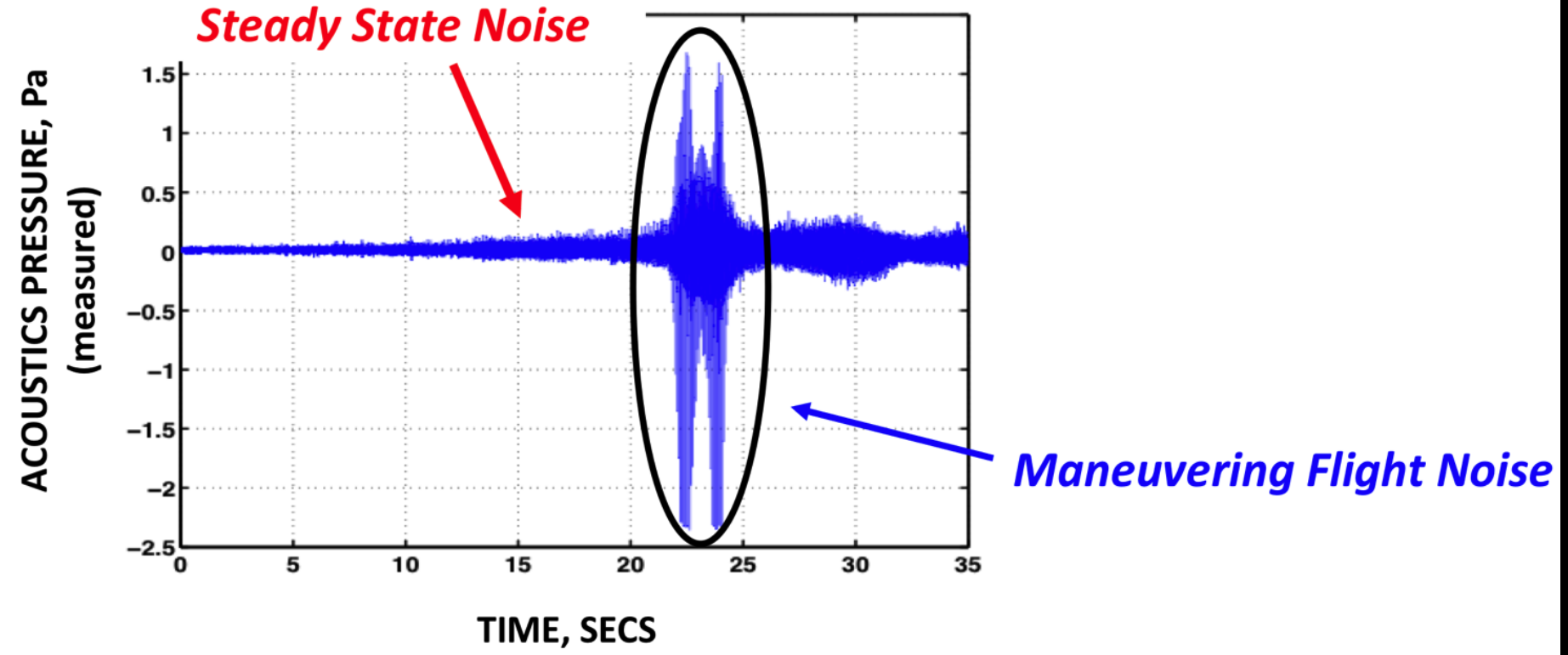
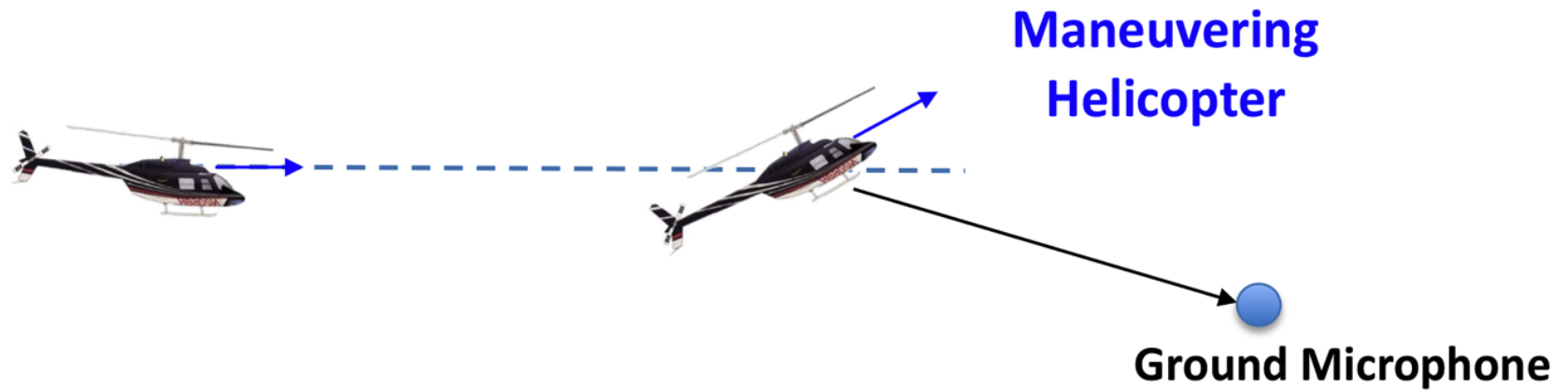
# 2007 NRTC Bell 206 Acoustic Flight Test







# Rapid maneuvers are often very noisy

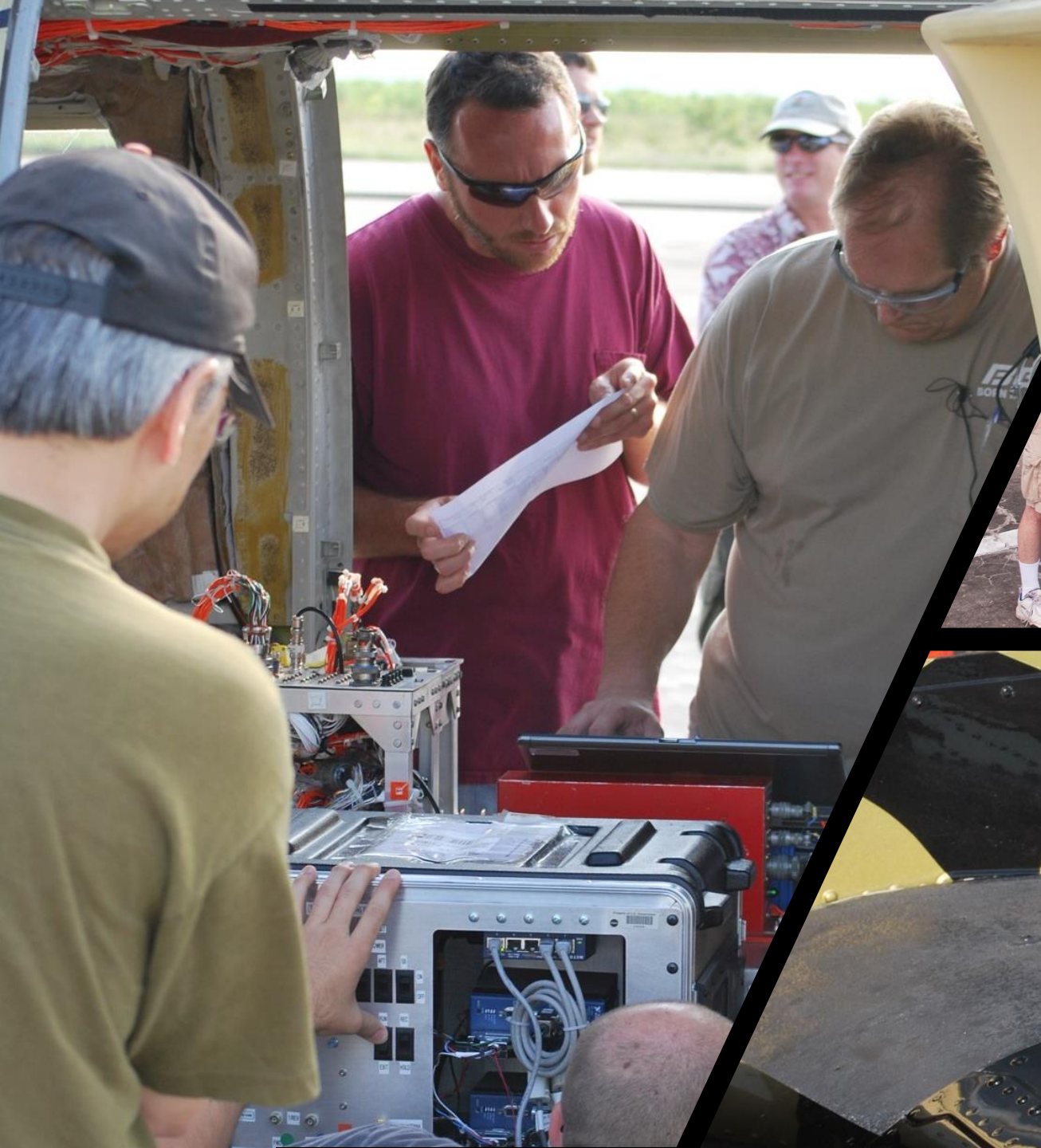




# 2011 NASA/Bell/Army Maneuver Noise Test







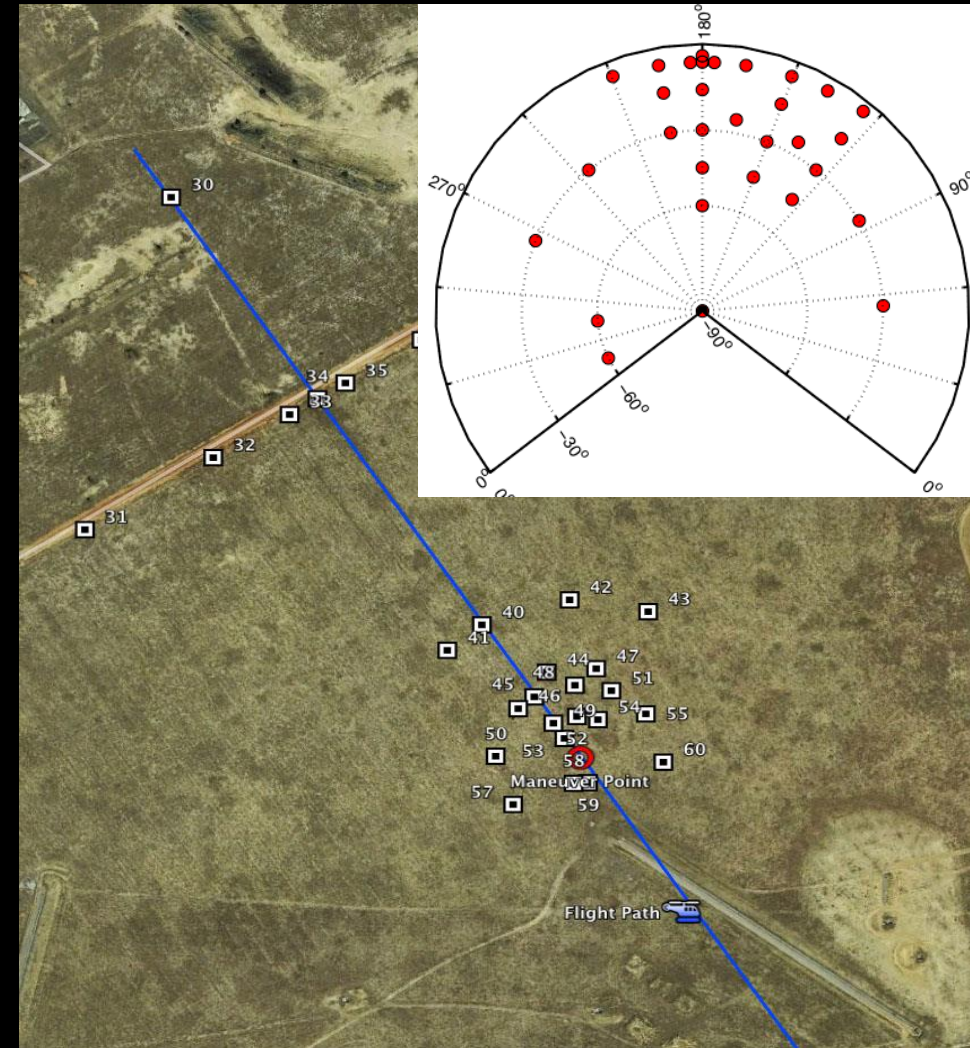
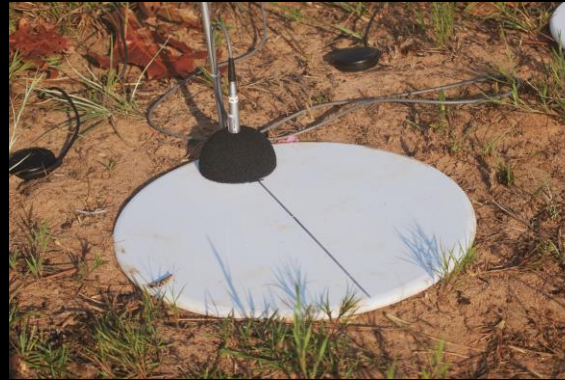




Command & Control Trailer



Microphone on Ground Board

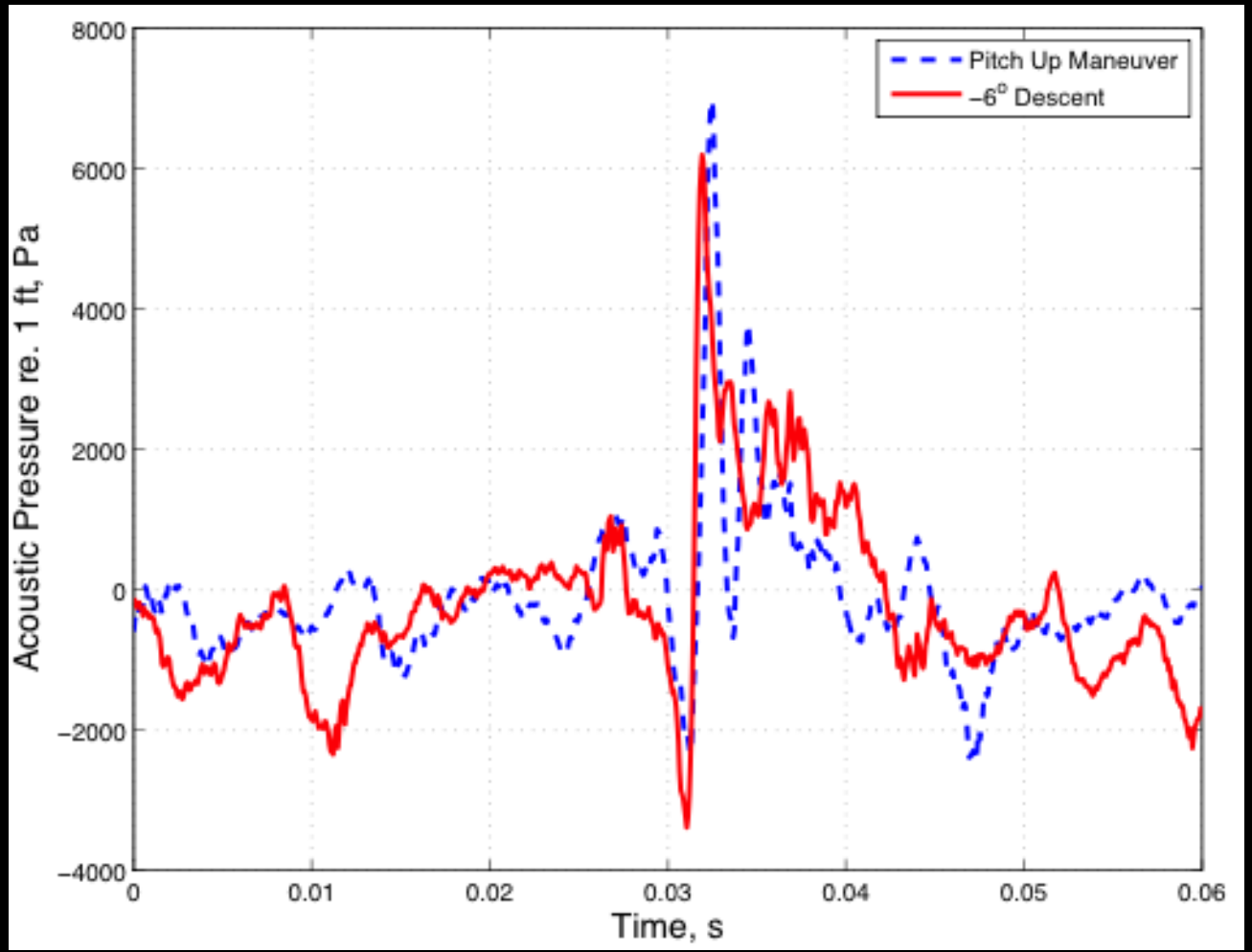


Eglin Maneuver Array





When you can look at it the right way, maneuvering noise doesn't seem different than steady flight

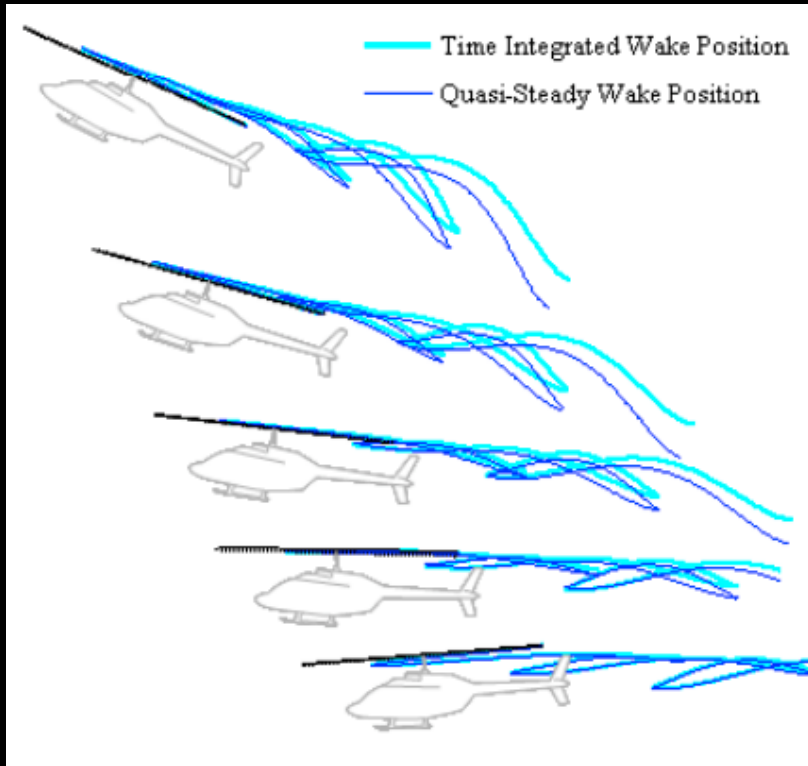




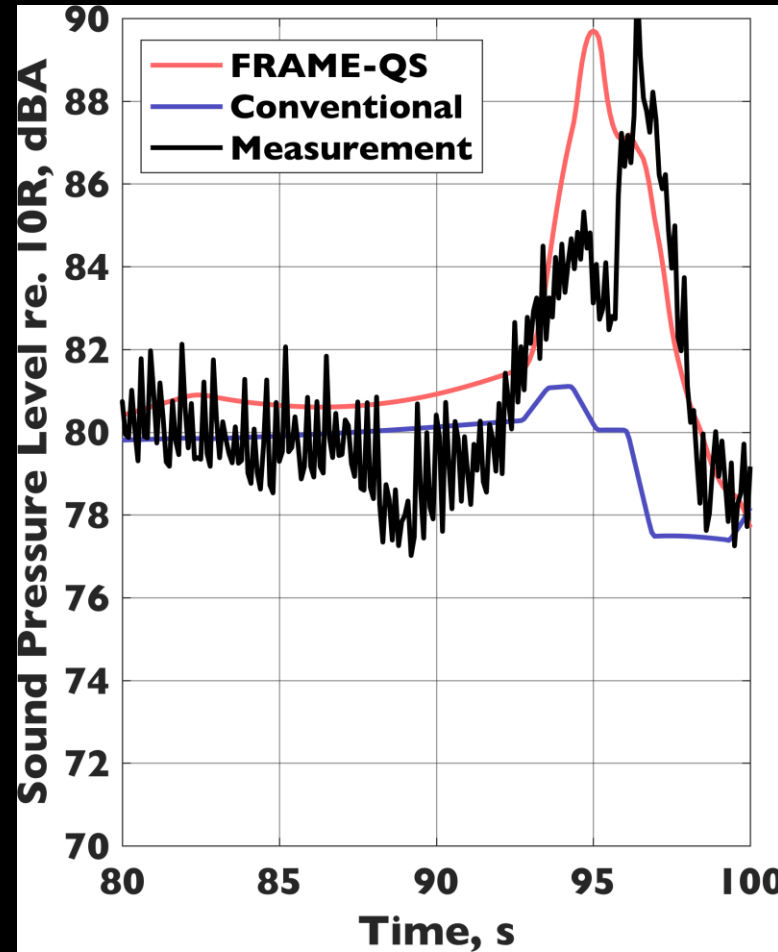
Can we use the steady flight noise data we normally collect to model maneuvers?



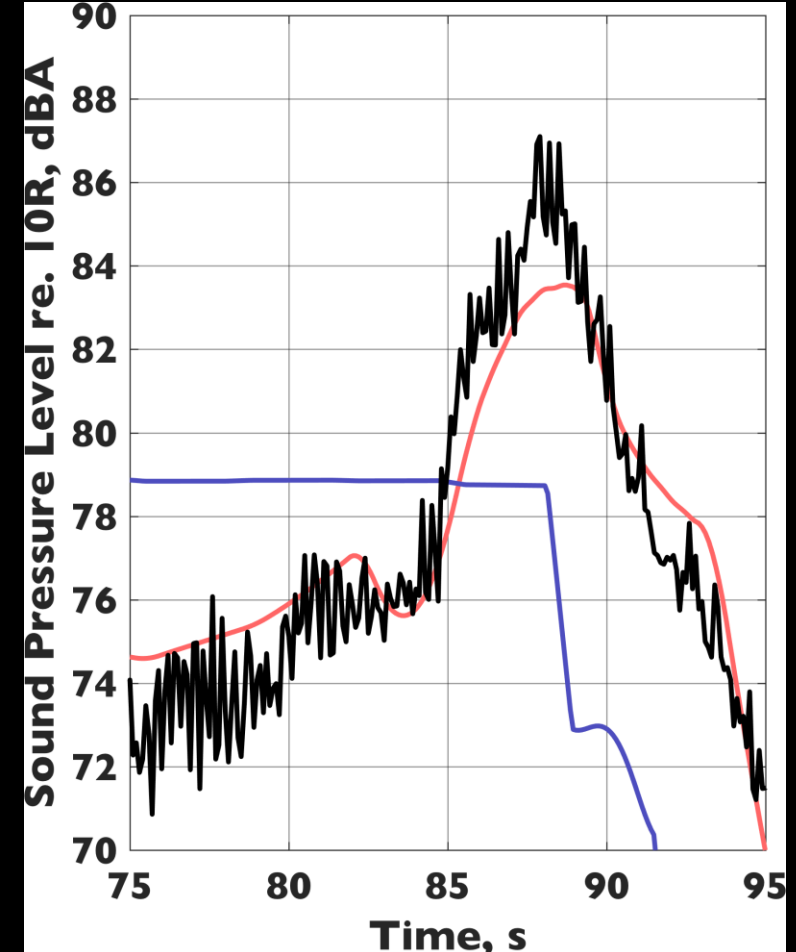
# Yes! By mapping the rotor aerodynamics to acoustics!



Sickenberger, R.D., Modeling Helicopter Near-Horizon Harmonic Noise due to Maneuvers, PhD Dissertation, University of Maryland, May 2013.



Pitch Up



Advancing Side Roll



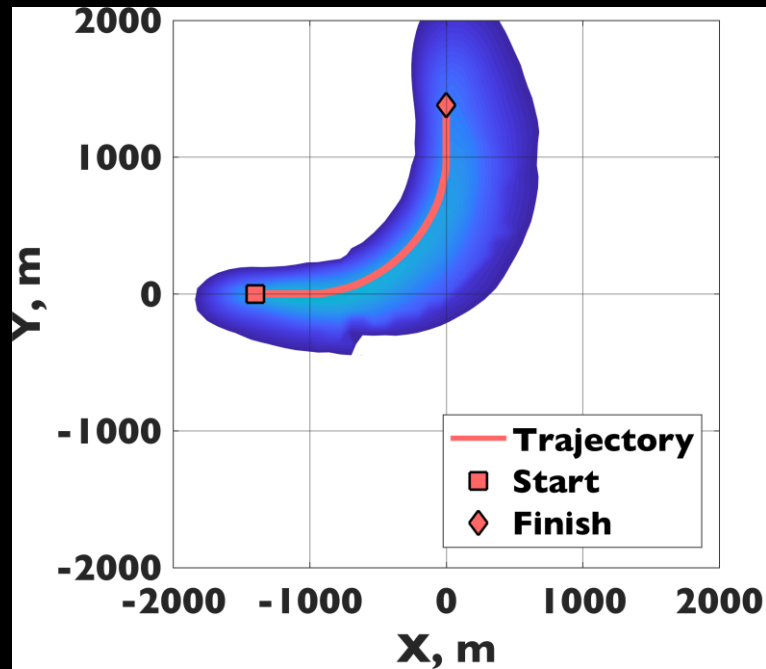
Can we use our models to develop  
useful guidance for pilots?



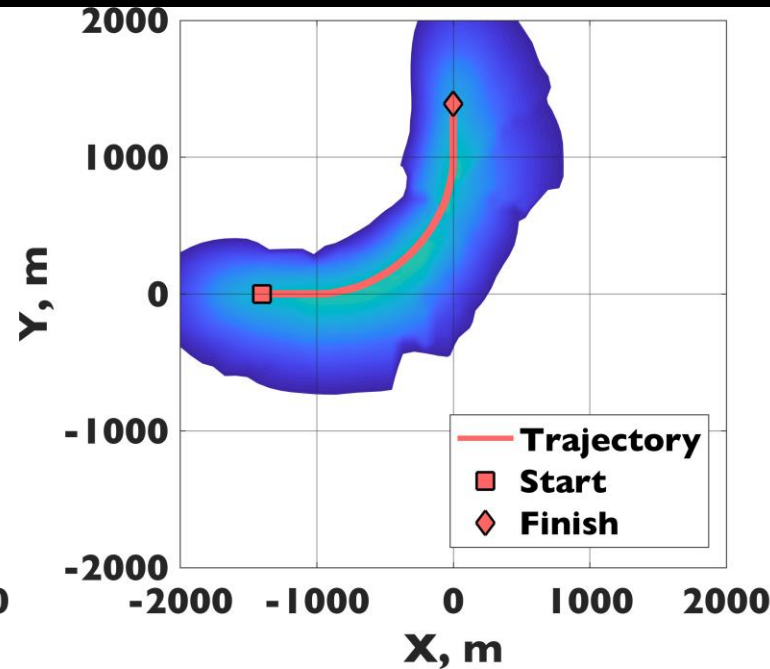


# Descending turns can be very loud!

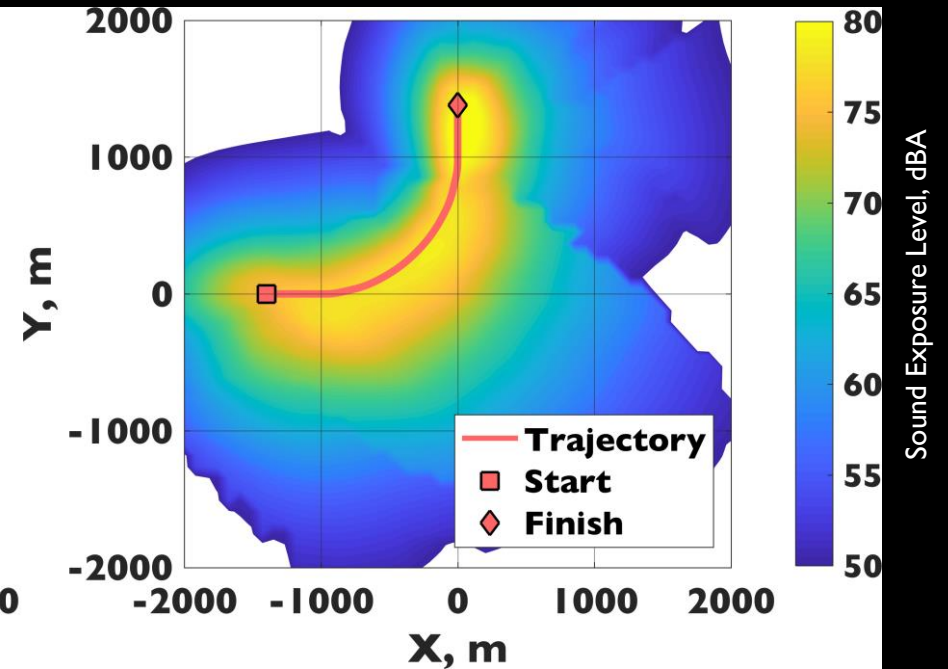
6° climbing turn



Level turn



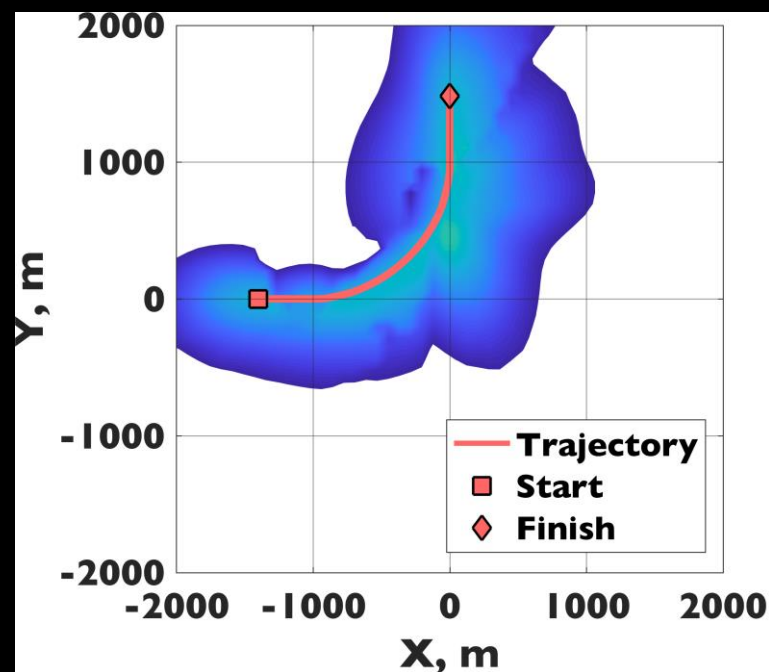
6° descending turn



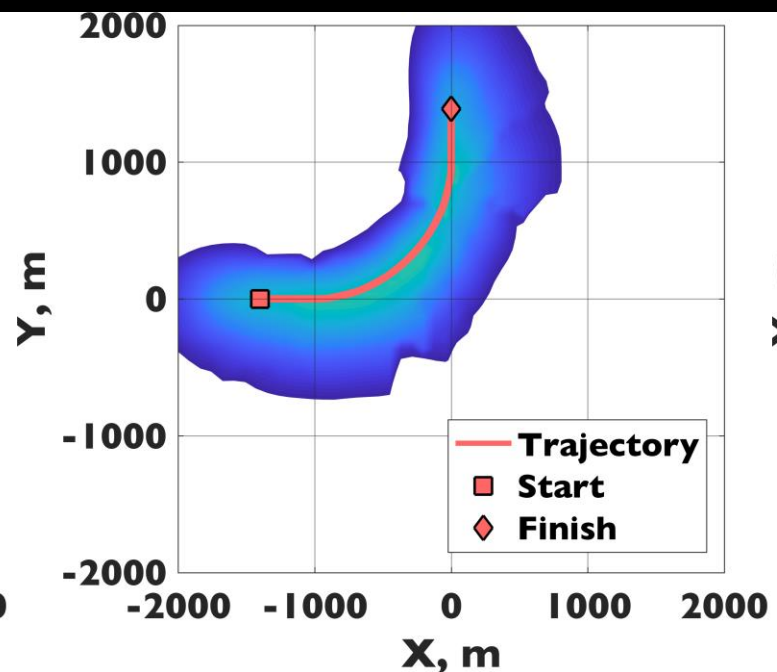


# Deceleration also tends to increase noise.

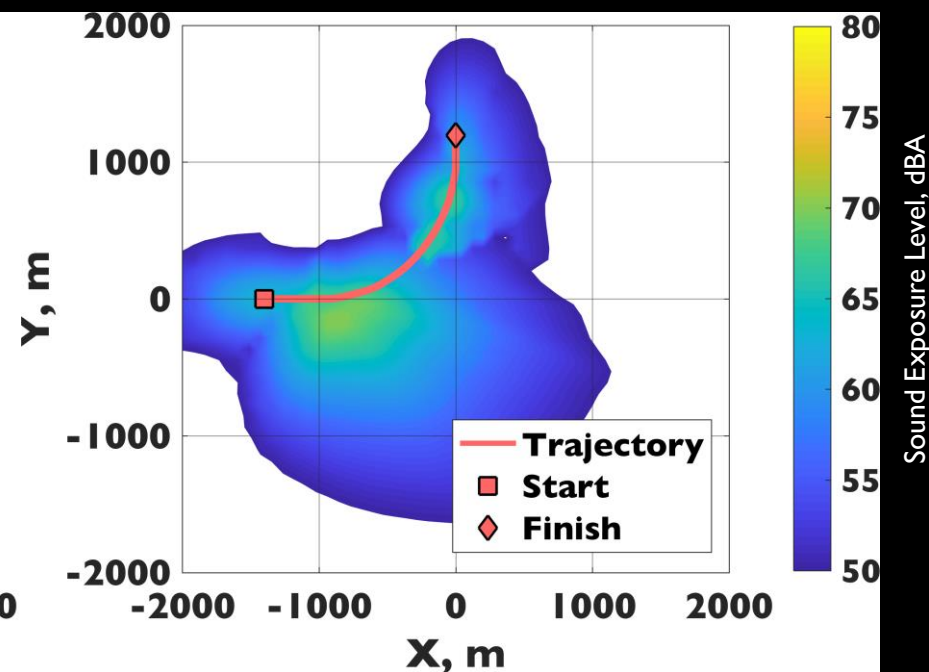
0.05 g accel



Steady



0.05 g decel





# Noise Abatement Guidance

- Avoid tendency to decelerate during maneuvers, such as turns or pull-ups
- Sustained acceleration leads to high noise levels; however, a little acceleration can add margin to BVI onset during noise sensitive maneuvers
- Keep noise sensitive areas inside of turns
- Avoid descending during turns toward the advancing side of the rotor
- Steeper climbs reduce noise on the ground, so long as the pull-up into climb is gradual

But, is this good advice for all helicopters?



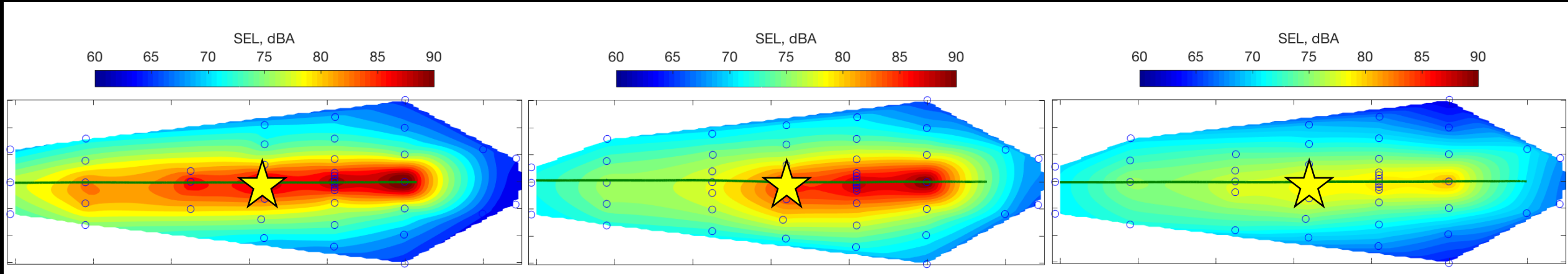
# 2017 NASA / FAA Noise Abatement Test







# R44 Noise Abatement Approaches



## Least Effective Approach Flown Shallow Decelerating Approach

-3.5° Flight Path Angle  
80 KIAS – 20 KIAS, 0.05g decel

100.3 EPNdB  
97.2 dBA SEL

## Prior Noise Abatement Guidance Normal Decelerating Approach

-4.5° Flight Path Angle  
80 KIAS – 20 KIAS, 0.05g decel

96.3 EPNdB  
93.1 dBA SEL

## Best Approach Flown Steep Approach to Flare

-6° Flight Path Angle  
80 KIAS until 0.15g flare to 20 KIAS

90.2 EPNdB  
85.4 dBA SEL

10.3 EPNdB / 12.0 dBA SEL benefit re. Least Effective  
6.3 EPNdB / 8.0 dBA SEL benefit re. Historical Guidance



# R44 Maneuvers

Level Turn

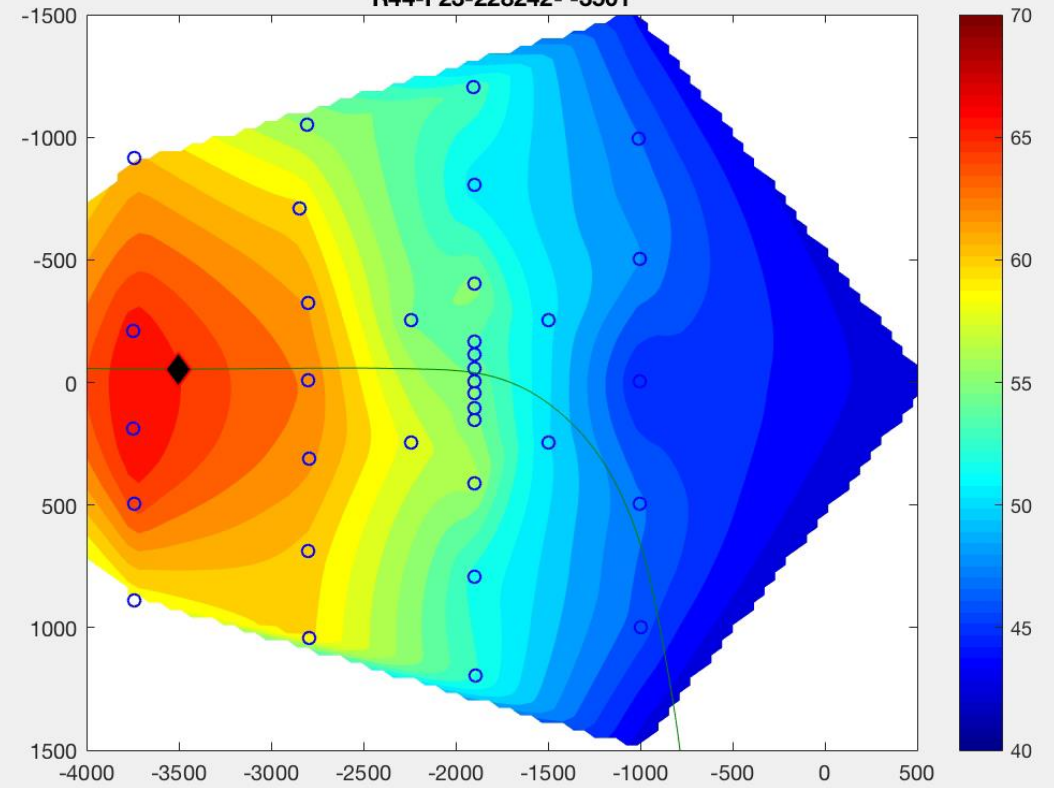
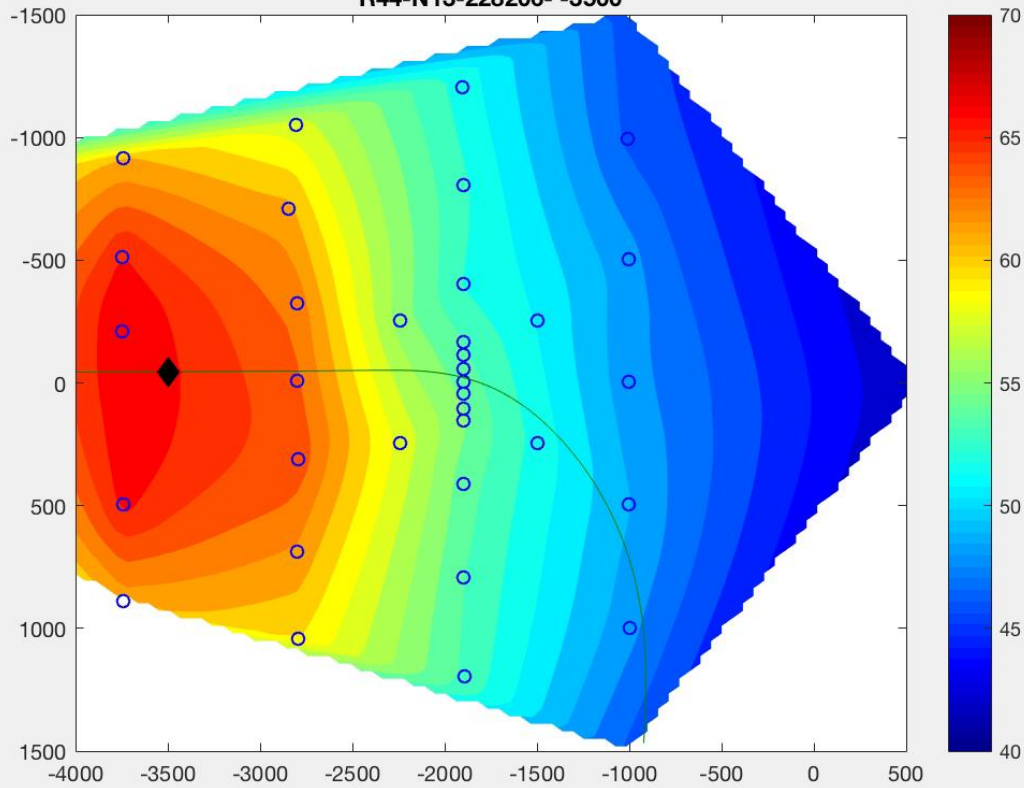
Descending Turn

dBA

R44-N13-228206 - -3500

R44-F23-228242 - -3501

Y, feet



X, feet

X, feet

2019 NASA /  
FAA Medium  
Helicopter  
Noise  
Abatement Test

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# Noise Abatement Guidelines

- General-purpose rules of thumb
- Validated for all ten aircraft
- Now included as a part of HAI's Fly Neighborly Curriculum
- Feedback from early adopters indicates that it's working!



## Helicopter Noise Abatement Recommendations

### Level Flight:

- ✈ Accelerations are quieter than decelerations
- ✈ Straight flight is quieter than turning flight

### Turning Flight:

- ✈ Turning away from the advancing blade (especially when decelerating) is quieter than turning into the advancing blade
- ✈ Level turns are quieter than descending turns

### Descending Flight:

- ✈ Straight-in flight is quieter than turning flight
- ✈ Steeper approaches are quieter than shallow approaches

### Decelerations:

- ✈ Level flight decelerations are quieter than descending or turning flight decelerations

### Maneuvering:

- ✈ Smooth and gentle control inputs are quieter than rapid control inputs

These recommendations are flight tested and scientifically vetted by the U.S. Department of Transportation and NASA to support the Fly Neighborly Goals.

Take the Fly Neighborly training at: <https://go.usa.gov/xQPCW>

*Fly neighborly procedures/recommendations should be executed in the safest manner possible and followed only to the extent safety is not compromised.*



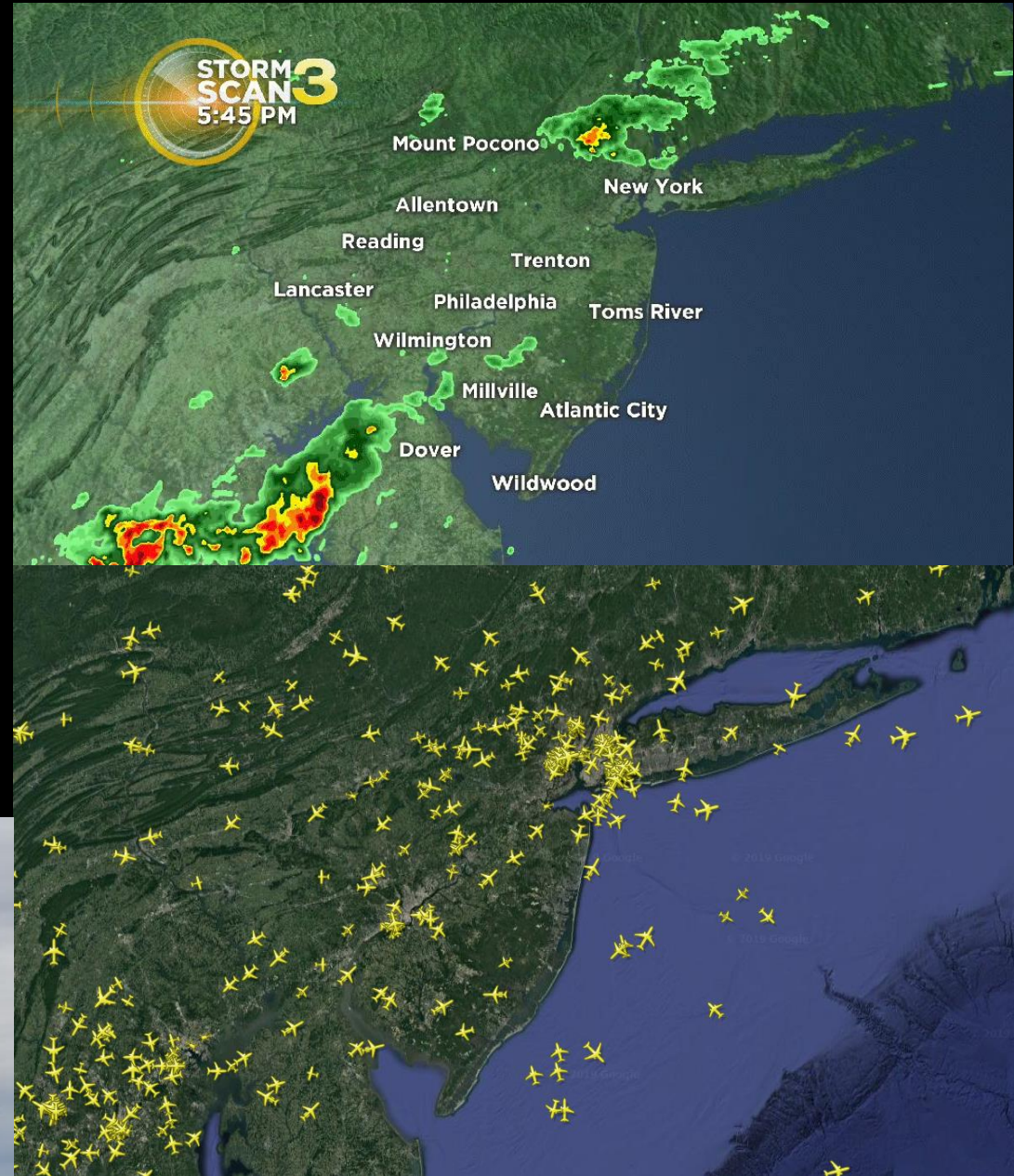




# The Best Laid Plans...

... may be obsolete tomorrow  
due to changes in:

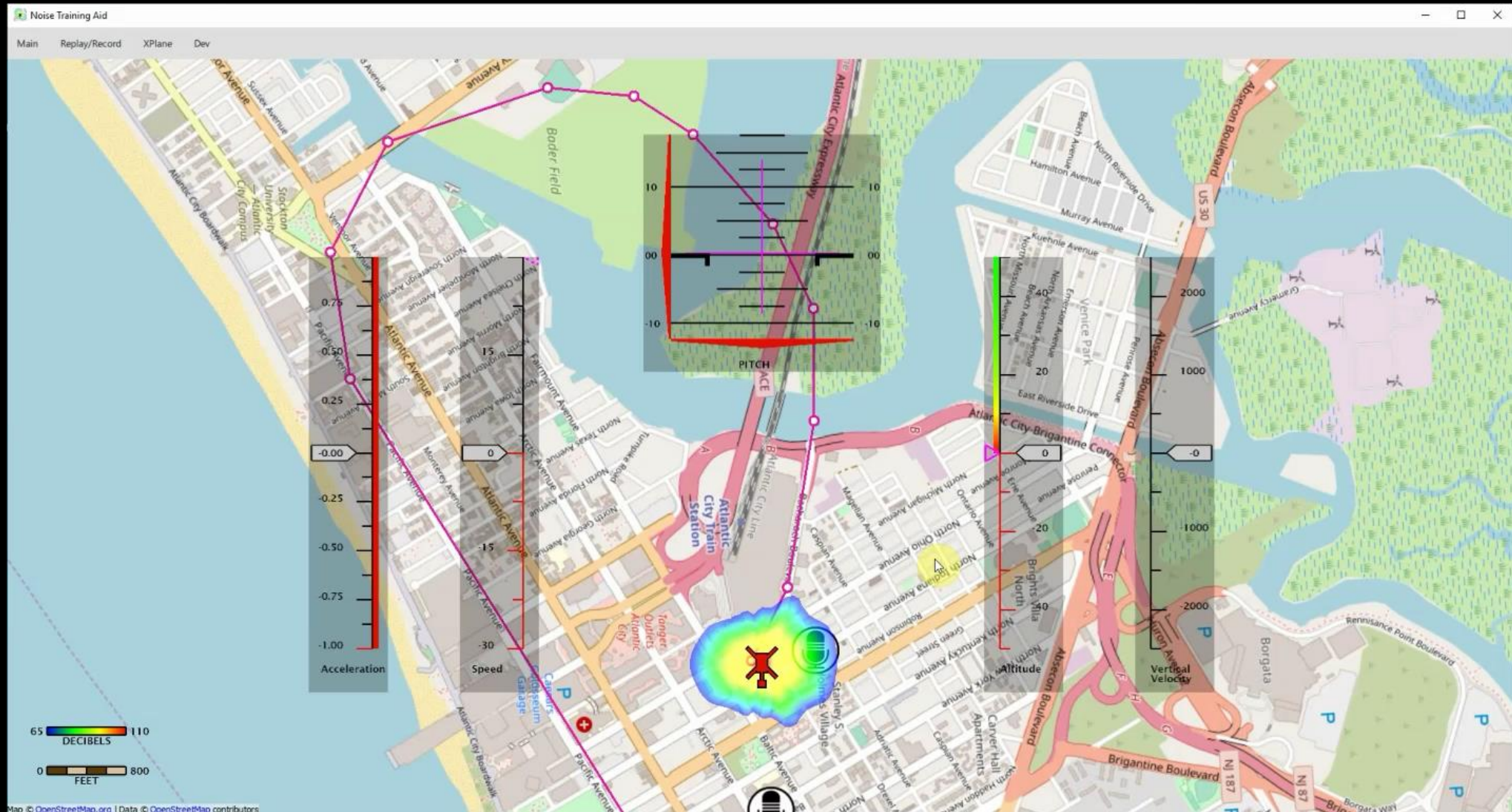
- Mission objectives
- Vehicle configuration
- Weather
- Air traffic





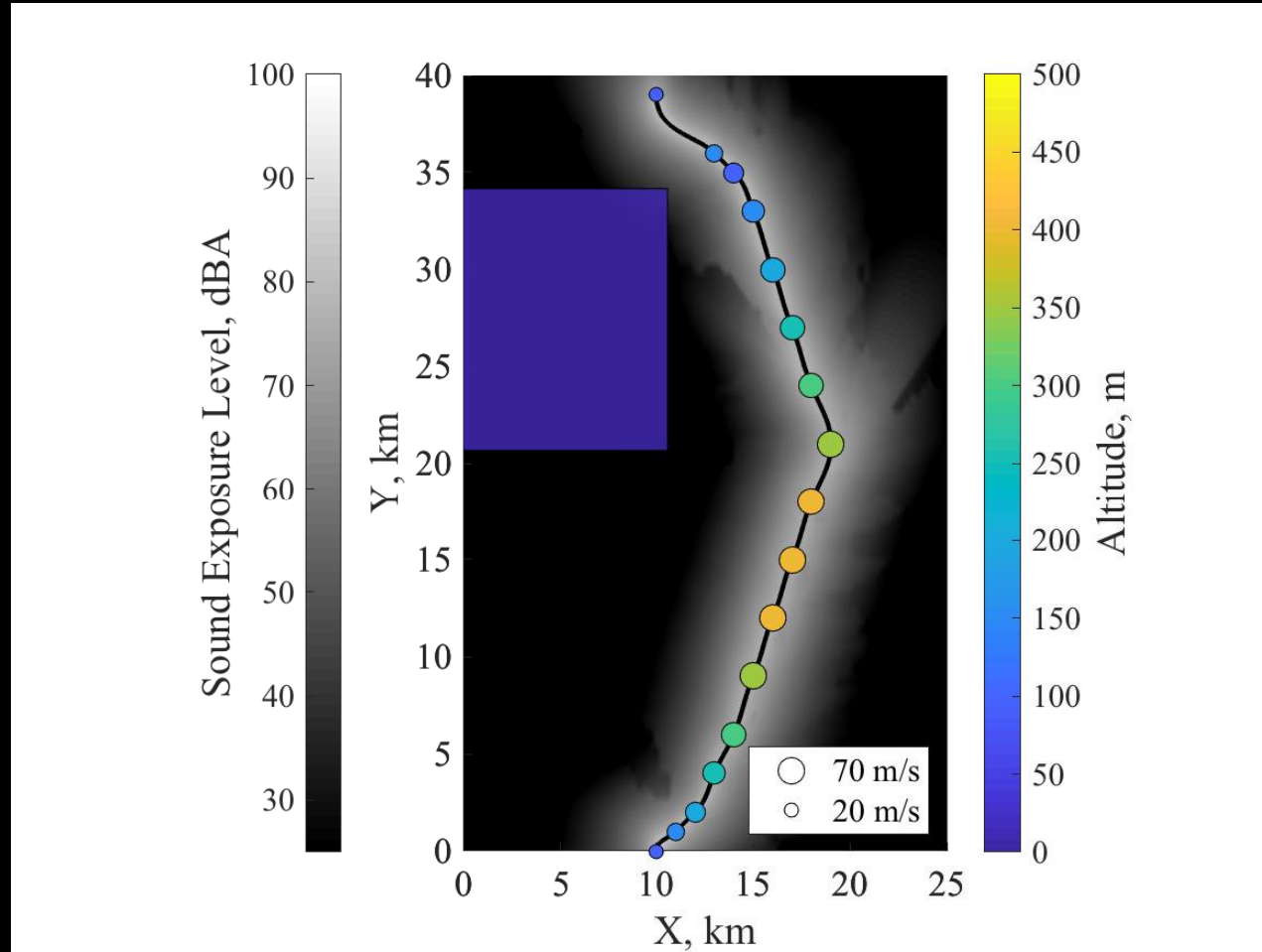


# Real Time Noise Awareness





# On Demand Mission Planning



## Area:

- Spans 20 x 40 km, similar area to Las Vegas, Rome, or Beijing

## Mission:

- Takeoff at Y = 0 km
- Land at Y = 40 km
- Closed airspace from 0 -  $\infty$  m

## Trajectory:

- About 20 minutes of flight time
- About 10 seconds computer time
- About one minute between nodes





# Dynamic Replanning







# eVTOL Noise

Rotor-Rotor  
Interaction  
(e.g. Boeing PAV)



Rotor-Wing  
Interaction  
(e.g. Cora)



Rotor-Stator  
Interaction  
(e.g. Bell Air Taxi)



- Electric motors don't help, the engines weren't the problem!
- Lower harmonic noise is less important with lower tip speeds
- Fluctuations in airloads cause loading noise at higher frequency harmonics:
  - Blade-Vortex Interaction (BVI)
  - Rotor-Rotor Interactions
  - Rotor-Wing Interactions
  - Rotor-Stator Interactions
- Rotor broadband will also be more significant:
  - Airfoil Self Noise
  - Blade-Wake Interaction



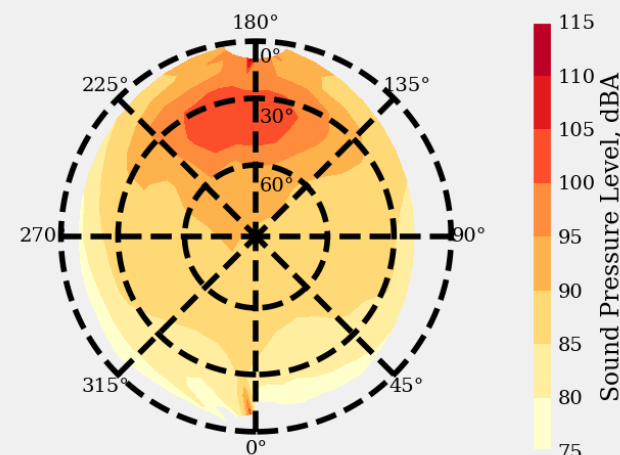
# Vehicle Configuration Control

- Most eVTOL have nonunique trim
- These nonunique trim states can be exploited for noise reduction
- “Acoustically-aware” flight controls should optimize vehicle configuration for best balance between noise exposure and performance

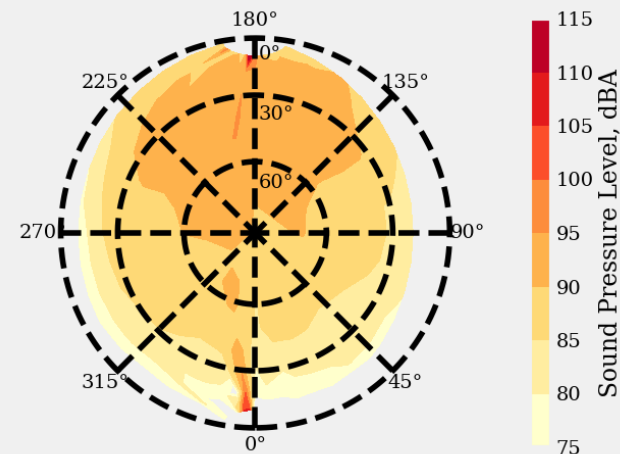


XV-15, 70 KIAS,  $-6^\circ$  FPA

Nacelle @  $80^\circ$



Nacelle @  $60^\circ$





# Concluding Remarks



- A decade of sustained research in low noise helicopter operations is yielding real reductions in community noise exposure
- Real-time noise modeling will soon provide operators with effective tailored guidance
- “Acoustically-aware” autonomy will eventually enable ultra-low-noise VTOL operations in and around communities





*Fin*

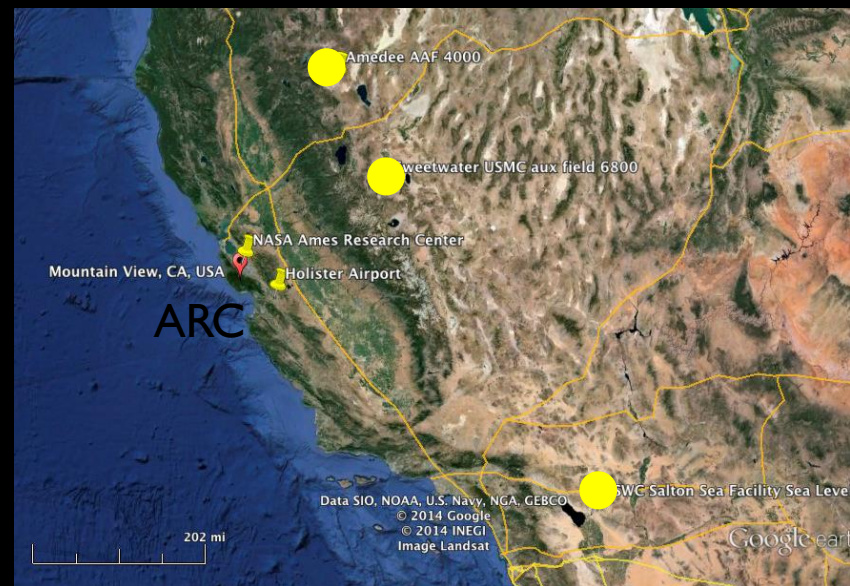
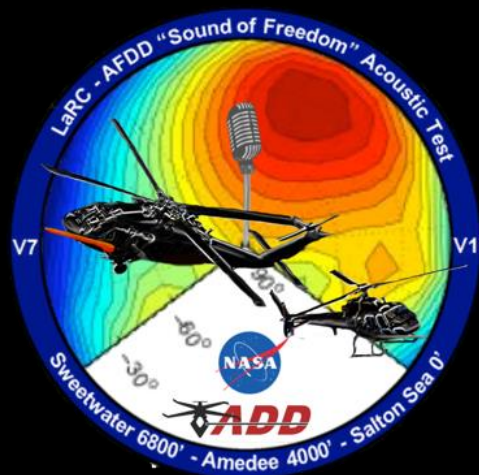


# 2016 NASA / Army Altitude Variation Test

## AS350



## EH-60L



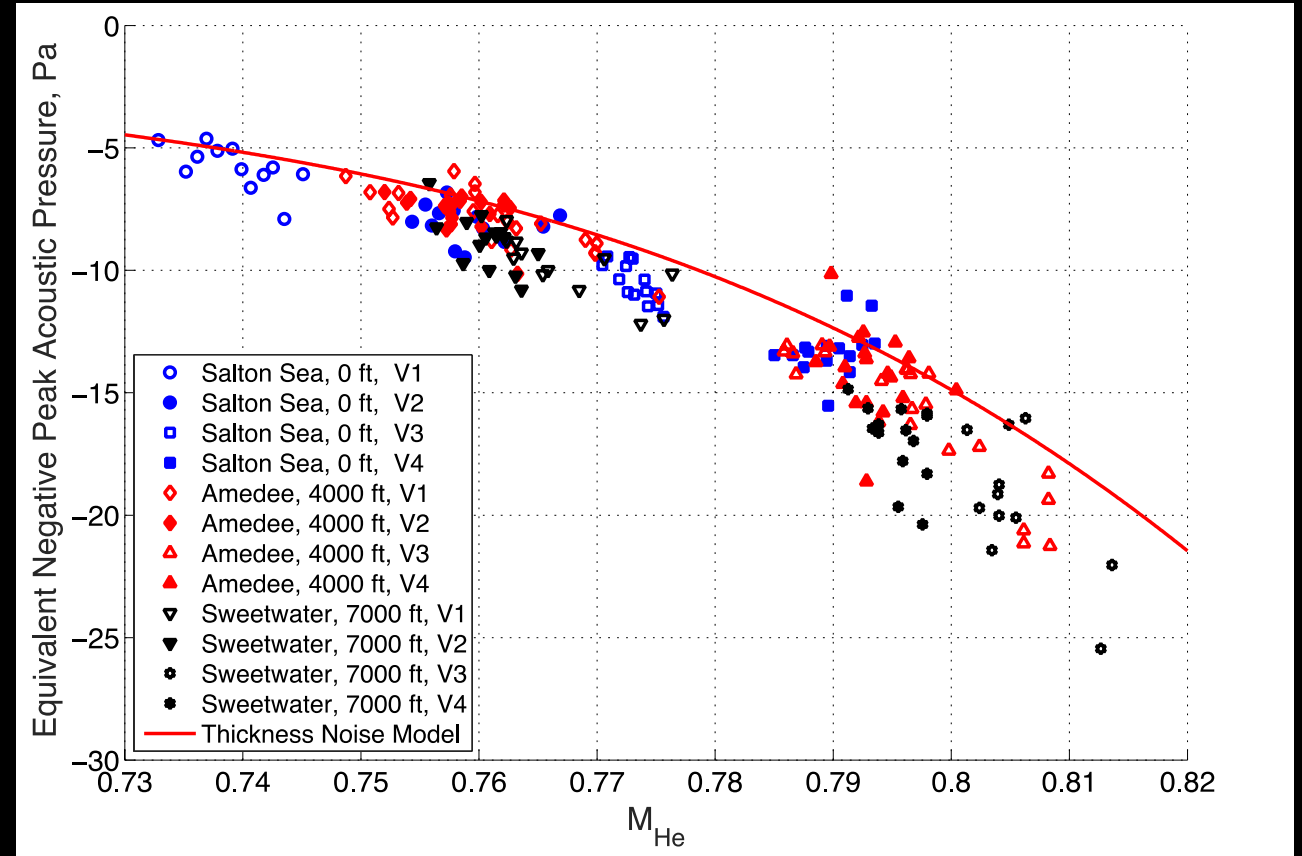
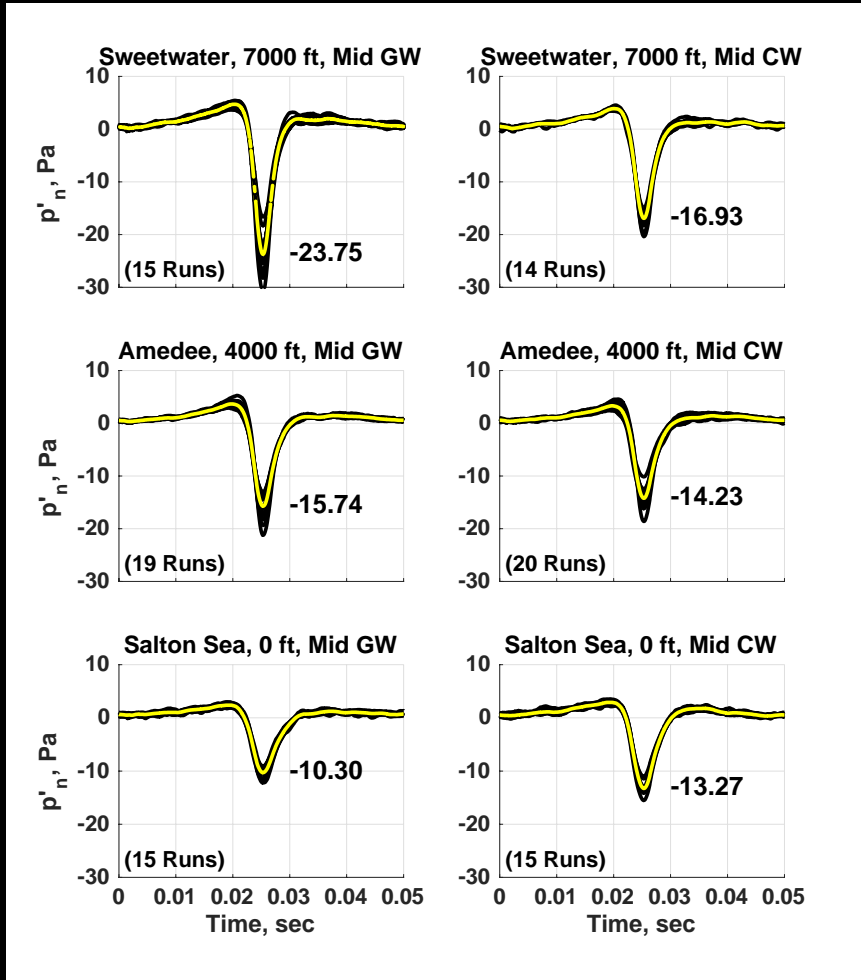
4000 feet  
Amedee Army Auxiliary  
Airfield

7000 feet  
USMC Mountain  
Warfare Training Center

Sea Level  
US Navy Salton Sea  
Facility



# Ambient conditions can change noise for the “same” condition by a factor of two!

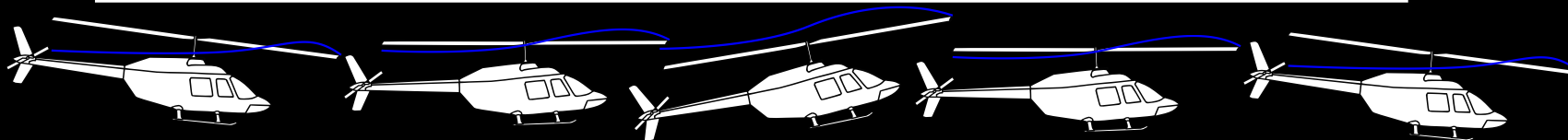
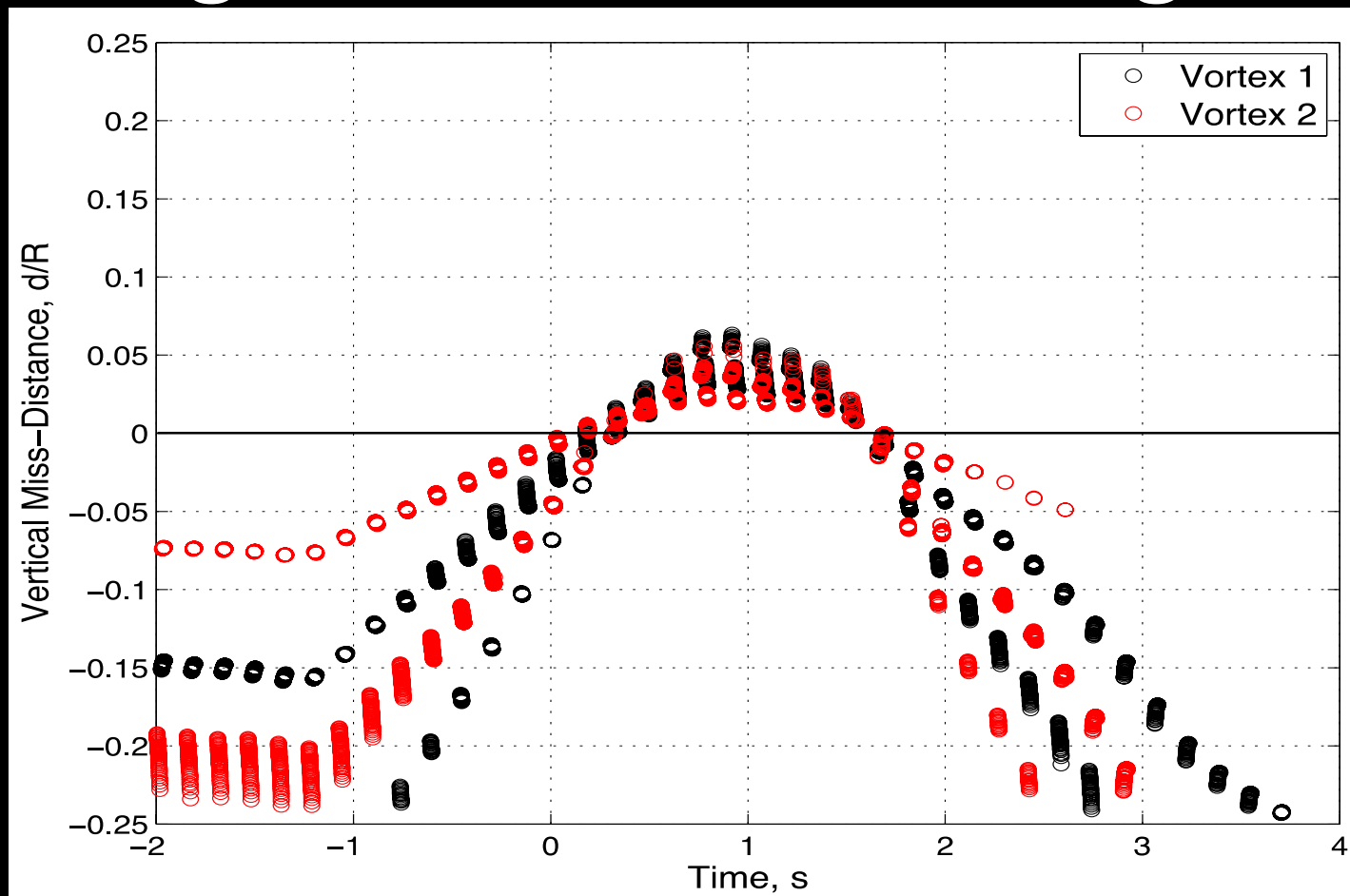


Nondimensionalization to the rescue!



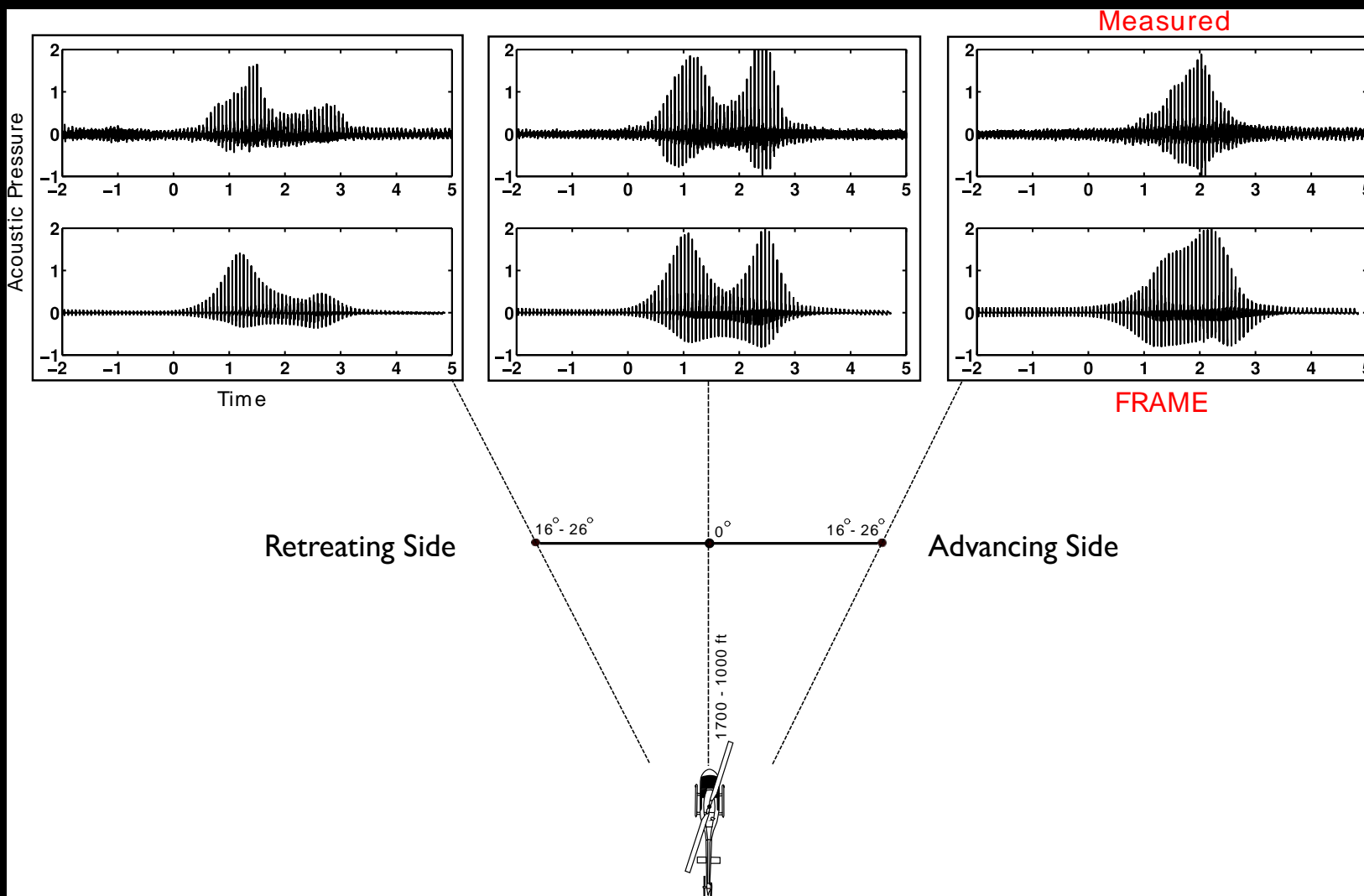


# Motion moves the wake up into the rotor... and then right back down through it





# But, with a good wake model, *we can* predict the noise



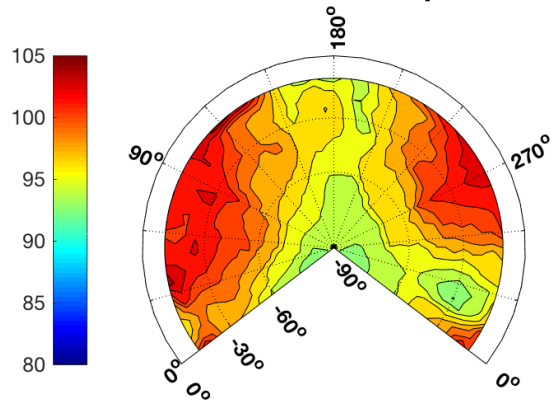


# Effects of Sideslip

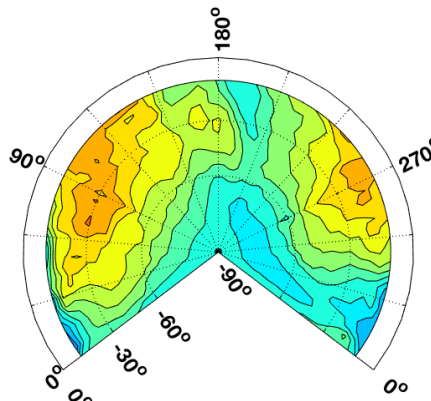
EC130B4

AS350B3

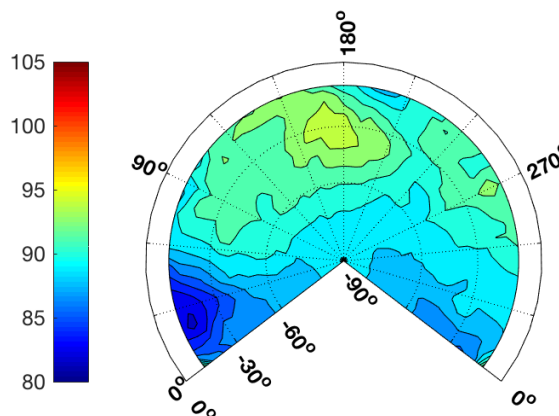
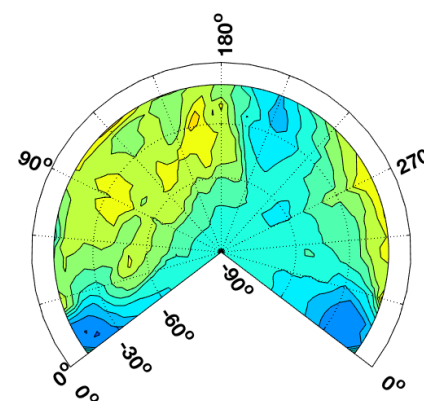
78.0 kts, -7.9 FPA  
left side slip



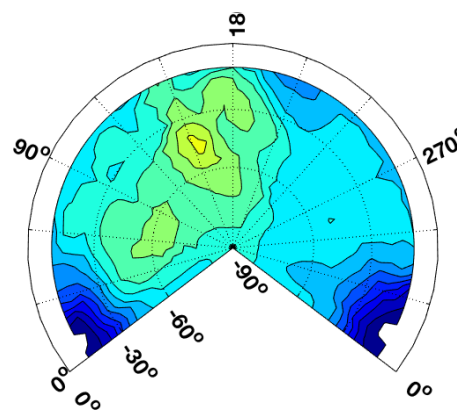
77.9 kts, -7.9 FPA



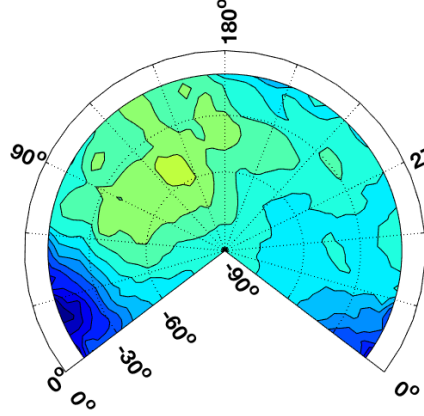
84.3 kts, -7.5 FPA  
right side slip



75.5 kts, -7.6 FPA  
left side slip



69.1 kts, -8.1 FPA



72.4 kts, -7.9 FPA  
right side slip